كنترل پيش بين **Model Predictive Control**

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MPC

- Basic Ideas of Predictive Control
- Receding Horizon Control Notation
- **OMPC** Features
- Stability and Invariance of MPC
- **Example 1** Feasibility and Stability
	- **O**Proof for $X_f = 0$
	- **O**General Terminal Sets
	- **O** Example
- Extension to Nonlinear MPC

Basic Ideas of Predictive Control

Infinite Time Constrained Optimal Control

(what we would like to solve)

$$
J_0^*(x(0)) = \min \sum_{k=0}^{\infty} q(x_k, u_k)
$$

s.t. $x_{k+1} = Ax_k + Bu_k, k = 0, ..., N - 1$
 $x_k \in \mathcal{X}, u_k \in \mathcal{U}, k = 0, ..., N - 1$
 $x_0 = x(0)$

 \Box Stage cost q(x, u) describes "cost" of being in state x and applying input u

- Optimizing over a trajectory provides a tradeoff between short- and long-term benefits of actions
- We'll see that such a control law has many beneficial properties.... but we can't compute it: there are an infinite number of variables

Basic Ideas of Predictive Control

Receding Horizon Control (what we can sometimes solve)

$$
J_t^*(x(t)) = \min_{U_t} \qquad p(x_{t+N}) + \sum_{k=0}^{N-1} q(x_{t+k}, u_{t+k})
$$

subj. to $x_{t+k+1} = Ax_{t+k} + Bu_{t+k}, k = 0, ..., N-1$
 $x_{t+k} \in \mathcal{X}, u_{t+k} \in \mathcal{U}, k = 0, ..., N-1$
 $x_{t+N} \in \mathcal{X}_f$
 $x_t = x(t)$ (1)

where $U_t = \{u_t, ..., u_{t+N-1}\}.$

Truncate after a finite horizon:

- $p(x_{t+N})$: Approximates the 'tail' of the cost
- \blacksquare \mathcal{X}_f : Approximates the 'tail' of the constraints

Basic Ideas of Predictive Control

On-line Receding Horizon Control

- 1) MEASURE the state $x(t)$ at time instance t
- 2) OBTAIN $U_t^*(x(t))$ by solving the optimization problem in (1)
- 3) IF $U_t^*(x(t)) = \emptyset$ THEN 'problem infeasible' STOP
- 4) APPLY the first element u_t^* of U_t^* to the system
- 5) WAIT for the new sampling time $t + 1$, GOTO 1)

Note that, we need a constrained optimization solver for step 2).

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RHC Notation

$$
x(t+1) = Ax(t) + Bu(t)
$$

\n
$$
y(t) = Cx(t)
$$

\n
$$
x(t) \in \mathcal{X}, \ u(t) \in \mathcal{U}, \ \forall t \ge 0
$$

The CFTOC Problem

$$
J_t^*(x(t)) = \min_{U_{t \to t+N|t}} p(x_{t+N|t}) + \sum_{k=0}^{N-1} q(x_{t+k|t}, u_{t+k|t})
$$

subject to $x_{t+k+1|t} = Ax_{t+k|t} + Bu_{t+k|t}, k = 0, ..., N-1$
 $x_{t+k|t} \in \mathcal{X}, u_{t+k|t} \in \mathcal{U}, k = 0, ..., N-1$
 $x_{t+N|t} \in \mathcal{X}_f$
 $x_{t|t} = x(t)$

with $U_{t\to t+N|t} = \{u_{t|t}, \ldots, u_{t+N-1|t}\}.$

RHC Notation

- \blacksquare $x(t)$ is the state of the system at time t.
- $x_{t+k|t}$ is the state of the model at time $t+k$, predicted at time t obtained by starting from the current state $x_{t|t} = x(t)$ and applying to the system model

$$
x_{t+1|t} = Ax_{t|t} + Bu_{t|t}
$$

the input sequence $u_{t|t}, \ldots, u_{t+k-1|t}$.

- For instance, $x_{3|1}$ represents the predicted state at time 3 when the prediction is done at time $t = 1$ starting from the current state $x(1)$. It is different, in general, from $x_{3|2}$ which is the predicted state at time 3 when the prediction is done at time $t = 2$ starting from the current state $x(2)$.
- Similarly $u_{t+k|t}$ is read as "the input u at time $t+k$ computed at time t".

RHC Notation

■ Let $U^*_{t\to t+N|t} = \{u^*_{t|t}, \ldots, u^*_{t+N-1|t}\}\$ be the optimal solution. The first element of $U^*_{t\to t+N|t}$ is applied to system

 $u(t) = u_{t|t}^*(x(t)).$

■ The CFTOC problem is reformulated and solved at time $t + 1$, based on the new state $x_{t+1|t+1} = x(t+1)$.

Receding horizon control law

$$
f_t(x(t)) = u_{t|t}^*(x(t))
$$

Closed loop system

$$
x(t+1) = Ax(t) + Bf_t(x(t)) \triangleq f_{cl}(x(t)), \ t \ge 0
$$

RHC Notation: Time-invariant Systems

As the system, the constraints and the cost function are time-invariant, the solution $f_t(x(t))$ becomes a time-invariant function of the initial state $x(t)$. Thus, we can simplify the notation as

 $\mathbf{A} \mathbf{F}$ \mathbf{H}

$$
J_0^*(x(t)) = \min_{U_0} p(x_N) + \sum_{k=0}^{N-1} q(x_k, u_k)
$$

subj. to

$$
x_{k+1} = Ax_k + Bu_k, \ k = 0, ..., N-1
$$

$$
x_k \in \mathcal{X}, \ u_k \in \mathcal{U}, \ k = 0, ..., N-1
$$

$$
x_N \in \mathcal{X}_f
$$

$$
x_0 = x(t)
$$

where $U_0 = \{u_0, \ldots, u_{N-1}\}.$

The control law and closed loop system are time-invariant as well, and we write $f_0(x_0)$ for $f_t(x(t))$.

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OPros

- Any model
	- **□** linear
	- **<u>O**</u> nonlinear
	- single/multivariable
	- **<u>I</u>** time delays
	- \Box constraints
- **Any objective:**
	- sum of squared errors
	- sum of absolute errors (i.e.,integral)
	- worst error over time
	- **exercise** conomic objective

O Cons

- **Q**Computationally demanding in the general case
- May or may not be stable
- May or may not be feasibles

Example: Cessna Citation Aircraft
Linearized continuous-time model:

(at altitude of 5000m and a speed of 128.2 m/sec)

$$
\dot{x} = \begin{bmatrix} -1.2822 & 0 & 0.98 & 0 \\ 0 & 0 & 1 & 0 \\ -5.4293 & 0 & -1.8366 & 0 \\ -128.2 & 128.2 & 0 & 0 \end{bmatrix} x + \begin{bmatrix} -0.3 \\ 0 \\ -17 \\ 0 \end{bmatrix} u
$$
\n
$$
y = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} x
$$
\n
$$
y = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} x
$$

- \blacksquare Input: elevator angle
- States: x_1 : angle of attack, x_2 : pitch angle, x_3 : pitch rate, x_4 : altitude
- Outputs: pitch angle and altitude
- Constraints: elevator angle ± 0.262 rad ($\pm 15^{\circ}$), elevator rate ± 0.524 rad $(\pm 60^{\circ})$, pitch angle ± 0.349 $(\pm 39^{\circ})$

Open-loop response is unstable (open-loop poles: 0, 0, $-1.5594 \pm 2.29i$)

LQR and Linear MPC with Quadratic Cost

- Quadratic cost
- Linear system dynamics
- Linear constraints on inputs and states

LQR
\n
$$
J_{\infty}(x(t)) = \min \sum_{k=0}^{\infty} x_t^T Q x_t + u_k^T R u_k
$$
\n
$$
J_0^*(x(t)) = \min_{U_0} \sum_{k=0}^{N-1} x_k^T Q x_k + u_k^T R u_k
$$
\n
$$
s.t. \ x_{k+1} = A x_k + B u_k
$$
\n
$$
x_0 = x(t)
$$
\n
$$
x_0 = x(t)
$$

Assume: $Q = Q^T \succeq 0$, $R = R^T \succ 0$

Example: LQR with saturation

Linear quadratic regulator with saturated inputs.

At time $t = 0$ the plane is flying with a deviation of 10m of the desired altitude, i.e. $x_0 = [0; 0; 0; 10]$

Problem parameters: Sampling time 0.25sec, $Q = I, R = 10$

- Closed-loop system is unstable
- Applying LQR control and saturating the controller can lead to instability!

16

17

Example: MPC with Bound Constraints on Inputs

MPC controller with input constraints $|u_i| \leq 0.262$

Problem parameters: Sampling time 0.25sec, $Q = I$, $R = 10$, $N = 10$

 \Rightarrow System does not converge to desired steady-state but to a limit cycle

18

Example: MPC with all Input Constraints

MPC controller with input constraints $|u_i| \leq 0.262$ and rate constraints $|\dot{u}_i| \leq 0.349$ approximated by $|u_k - u_{k-1}| \leq 0.349 T_s$

Problem parameters: Sampling time 0.25sec, $Q = I$, $R = 10$, $N = 10$

- The MPC controller considers all constraints on the actuator
	- Closed-loop system is stable
	- \blacksquare Efficient use of the control authority

19

Example: Inclusion of state constraints

MPC controller with input constraints $|u_i| \leq 0.262$ and rate constraints $|u_i| \leq 0.349$ approximated by $|u_k - u_{k-1}| \leq 0.349 T_s$

Problem parameters: Sampling time 0.25sec, $Q=I,\;R=10,\;N=10$

Increase step:

At time $t = 0$ the plane is flying with a deviation of 100m of the desired altitude. i.e. $x_0 = [0; 0; 0; 100]$

Pitch angle too large during transient

Example: Short horizon

MPC controller with input constraints $|u_i| \leq 0.262$ and rate constraints $|u_i| \leq 0.349$ approximated by $|u_k - u_{k-1}| \leq 0.349 T_s$

Problem parameters: Sampling time 0.25sec, $Q = I$, $R = 10$, $N = 10$

Add state constraints for passenger comfort:

 $|x_2| \leq 0.349$

Example: Short horizon

MPC controller with input constraints $|u_i| \leq 0.262$ and rate constraints $|\dot{u}_i| \leq 0.349$ approximated by $|u_k - u_{k-1}| \leq 0.349 T_s$

20 .5 Altitude x_4 (m) 0.5 -20 10 $\overline{2}$ $\overline{0}$ 6 8 4 Time (sec) 0.5 Elevator angle u (rad) -0.5 ₀ $\overline{2}$ 6 8 10 Time (sec)

Problem parameters:

Sampling time 0.25sec, $Q = I, R = 10, N = 4$

Decrease in the prediction horizon causes loss of the stability properties

Pitch angle x₂ (rad)

21

Example: Short horizon

MPC controller with input constraints $|u_i| \leq 0.262$ and rate constraints $|\dot{u}_i| \leq 0.349$ approximated by $|u_k - u_{k-1}| \leq 0.349 T_s$

Problem parameters:

Sampling time 0.25sec, $Q = I, R = 10, N = 4$

Inclusion of terminal cost and constraint provides stability

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Loss of Feasibility and Stability

■ What can go wrong with "standard" MPC?

- No feasibility guarantee, i.e., the MPC problem may not have a solution
- No stability guarantee, i.e., trajectories may not converge to the origin

\Box Infeasibility can be due to:

- modeling errors
- disturbances
- wrong MPC setup (e.g., prediction horizon is too short)

Example : Loss of feasibility - Double Integrator

Consider the double integrator

$$
\begin{cases}\nx(t+1) = \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} x(t) + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t) \\
y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} x(t)\n\end{cases}
$$

subject to the input constraints

$$
-0.5 \le u(t) \le 0.5
$$

and the state constraints

$$
\begin{bmatrix} -5 \\ -5 \end{bmatrix} \le x(t) \le \begin{bmatrix} 5 \\ 5 \end{bmatrix}.
$$

Compute a receding horizon controller with quadratic objective with

$$
N = 3, \ P = Q = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}, \ R = 10.
$$

25

 $1.00.050$

Example : Loss of feasibility - Double Integrator

The QP problem associated with the RHC is

 $P_1 - P_2 + P_3$

$$
H = \begin{bmatrix} 13.50 & -10.00 & -0.50 \\ -10.00 & 22.00 & -10.00 \\ -0.50 & -10.00 & 31.50 \end{bmatrix}, \ F = \begin{bmatrix} -10.50 & 10.00 & -0.50 \\ -20.50 & 10.00 & 9.50 \end{bmatrix}, \ Y = \begin{bmatrix} 14.50 & 23.50 \\ 23.50 & 54.50 \end{bmatrix}
$$

Example : Loss of feasibility - Double Integrator

- 1) MEASURE the state $x(t)$ at time instance t
- 2) OBTAIN $U_0^*(x(t))$ by solving the optimization problem in (1)
- 3) IF $U_0^*(x(t)) = \emptyset$ THEN 'problem infeasible' STOP
- 4) APPLY the first element u_0^* of U_0^* to the system
- 5) WAIT for the new sampling time $t + 1$, GOTO 1)

27

Depending on initial condition, closed loop trajectory may lead to states for which optimization problem is infeasible.

- trajectories
- \triangleright Go to mpcdoubleint.m in MPC Toolbox

Summary: Feasibility and Stability

Problems originate from the use of a 'short sighted' strategy

⇒ Finite horizon causes deviation between the open-loop prediction and the closed-loop system: Set of feasible

- \triangleright Ideally we would solve the MPC problem with an infinite horizon, but that is computationally intractable
- Design finite horizon problem such that it approximates the infinite horizon

Summary: Feasibility and Stability

Infinite-Horizon

If we solve the RHC problem for $N = \infty$ (as done for LQR), then the open loop trajectories are the same as the closed loop trajectories. Hence

- If problem is feasible, the closed loop trajectories will be always feasible
- If the cost is finite, then states and inputs will converge asymptotically to the origin
- Finite-Horizon

RHC is "short-sighted" strategy approximating infinite horizon controller. But

- Feasibility. After some steps the finite horizon optimal control problem may become infeasible. (Infeasibility occurs without disturbances and model mismatch!)
- Stability. The generated control inputs may not lead to trajectories that converge to the origin.

Feasibility and stability in MPC - Solution

• Main idea: Introduce terminal cost and constraints to explicitly ensure feasibility and stability:

$$
J_0^*(x_0) = \min_{U_0} \qquad p(x_N) + \sum_{k=0}^{N-1} q(x_k, u_k) \qquad \text{Terminal Cost}
$$
\n
$$
\text{subj. to}
$$
\n
$$
x_{k+1} = Ax_k + Bu_k, \ k = 0, \dots, N-1
$$
\n
$$
x_k \in \mathcal{X}, \ u_k \in \mathcal{U}, \ k = 0, \dots, N-1
$$
\n
$$
x_N \in \mathcal{X}_f
$$
\n
$$
x_0 = x(t)
$$

 $p(\cdot)$ and \mathcal{X}_f are chosen to mimic an infinite horizon.

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Feasibility and Stability of MPC: Proof

Main steps:

- Prove recursive feasibility by showing the existence of a feasible control sequence at all time instants when starting from a feasible initial point
- **Prove stability by showing that the optimal cost function is a Lyapunov function**

Two cases:

- 1. Terminal constraint at zero: $x_N = 0$
- 2. Terminal constraint in some (convex) set: $x_N \in X_f$

General notation:

$$
J_0^*(x_0) = \min_{U_0} \underbrace{p(x_N)}_{\text{terminal cost}} + \sum_{i=0}^{N-1} \underbrace{q(x_i, u_i)}_{\text{stage cost}}
$$

Quadratic case: $q(x_i, u_i) = x_i^T Q x_i + u_i^T R u_i$, $p(x_N) = x_N^T P x_N$

Stability of MPC - Zero terminal state constraint

Terminal constraint: $x_N \in \mathcal{X}_f = 0$

- Assume feasibility of x_0 and let $\{u_0^*, u_1^*, \ldots, u_{N-1}^*\}$ be the optimal control sequence computed at x_0 and $\{x(0), x_1, \ldots, x_N\}$ be the corresponding state trajectory
- Apply u_0^* and let system evolve to $x(1) = Ax_0 + Bu_0^*$
- At $x(1)$ the control sequence $\{u_1^*, u_2^*, \ldots, u_{N-1}^*, 0\}$ is feasible (apply 0 control input \Rightarrow $x_{N+1} = 0$)
- \Rightarrow Recursive feasibility \triangledown

 $\Rightarrow J_0^*(x)$ is a Lyapunov function \rightarrow (Lyapunov) Stability \triangledown

Stability of MPC - Zero terminal state constraint

Example: Impact of Horizon with Zero Terminal **Constraint**

System dynamics:

$$
x_{k+1} = \begin{bmatrix} 1.2 & 1 \\ 0 & 1 \end{bmatrix} x_k + \begin{bmatrix} 1 \\ 0.5 \end{bmatrix} u_k
$$

Constraints:

$$
\mathcal{X} := \{x \mid -50 \le x_1 \le 50, \ -10 \le x_2 \le 10\} = \{x \mid A_x x \le b_x\}
$$

$$
\mathcal{U} := \{u \mid ||u||_{\infty} \le 1\} = \{u \mid A_u u \le b_u\}
$$

Stage cost:

$$
q(x, u) := x' \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} x + u^T u
$$

36

Extension to More General Terminal Sets

Problem: The terminal constraint $x_N = 0$ reduces the size of the feasible set **Goal:** Use convex set X_f to increase the region of attraction

Goal: Generalize proof to the constraint $x_N \in X_f$

Invariant sets

Definition: Invariant set

A set O is called *positively invariant* for system $x(t + 1) = f_{cl}(x(t))$, if

 $x(0) \in \mathcal{O} \Rightarrow x(t) \in \mathcal{O}, \quad \forall t \in \mathbb{N}_+$

The positively invariant set that contains every closed positively invariant set is called the maximal positively invariant set \mathcal{O}_{∞} .

Stability of MPC - Main Result

Assumptions

- **1** Stage cost is positive definite, i.e. it is strictly positive and only zero at the origin
- Terminal set is **invariant** under the local control law $v(x_k)$: $\mathbf{2}$

$$
x_{k+1} = Ax_k + Bv(x_k) \in \mathcal{X}_f, \text{ for all } x_k \in \mathcal{X}_f
$$

All state and input constraints are satisfied in \mathcal{X}_f :

 $\mathcal{X}_f \subseteq \mathcal{X}, v(x_k) \in \mathcal{U}, \text{ for all } x_k \in \mathcal{X}_f$

Terminal cost is a continuous Lyapunov function in the terminal set \mathcal{X}_f and $3¹$ satisfies:

$$
p(x_{k+1}) - p(x_k) \leq -q(x_k, v(x_k)), \text{ for all } x_k \in \mathcal{X}_f
$$

Under those 3 assumptions:

Theorem

The closed-loop system under the MPC control law $u_0^*(x)$ is asymptotically stable and the set \mathcal{X}_f is positive invariant for the system $x(k+1) = Ax + Bu_0^*(x)$.

Stability of MPC - Outline of the Proof

Assume feasibility of $x(0)$ and let $\{u_0^*, u_1^*, \ldots, u_{N-1}^*\}$ be the optimal control sequence computed at $x(0)$ and $\{x(0), x_1, ..., x_N\}$
the corresponding state trajectory the corresponding state trajectory

\n- **A**t
$$
x(1)
$$
, $\{u_1^*, u_2^*, \ldots, v(x_N)\}$ is feasible:
\n- x_N is in $\mathcal{X}_f \to v(x_N)$ is feasible and $x_{N+1} = Ax_N + Bv(x_N)$ in \mathcal{X}_j
\n

 \Rightarrow Terminal constraint provides recursive feasibility

Stability of MPC - Outline of the Proof

- Assume feasibility of $x(0)$ and let
 $\{u_0^*, u_1^*, \ldots, u_{N-1}^*\}$ be the optimal control

sequence computed at $x(0)$ and $\{x(0), x_1, \ldots, x_N\}$

the corresponding state trajectory Assume feasibility of $x(0)$ and let the corresponding state trajectory
- At $x(1)$, $\{u_1^*, u_2^*, \ldots, v(x_N)\}\)$ is feasible: x_N is in $\mathcal{X}_f \to v(x_N)$ is feasible and $x_{N+1} = Ax_N + Bv(x_N)$ in \mathcal{X}_f

 \Rightarrow Terminal constraint provides recursive feasibility

Stability of MPC - Outline of the Proof

$$
J_0^*(x_0) = \sum_{i=0}^{N-1} q(x_i, u_i^*) + p(x_N)
$$

Feasible, sub-optimal sequence for $x_1 : \{u_1^*, u_2^*, \ldots, v(x_N)\}\$ $J_0^*(x_1) \le \sum_{i=1} q(x_i, u_i^*) + p(Ax_N + Bv(x_N))$ $N-1$ $= \sum_{i} q(x_i, u_i^*) + p(x_N) - q(x_0, u_0^*) + p(Ax_N + Bv(x_N))$ $- p(x_N) + q(x_N, v(x_N))$ = $J_0^*(x_0) - q(x_0, u_0^*) + p(Ax_N + Bv(x_N)) - p(x_N) + q(x_N, v(x_N))$ $p(x)$ < 0 $\implies J_0^*(x_1) - J_0^*(x_0) \leq -q(x_0, u_0^*), \quad q > 0$

 $J_0^*(x)$ is a Lyapunov function decreasing along the closed loop trajectories \Rightarrow The closed-loop system under the MPC control law is asymptotically stable

Feasibility and Stability

Choice of Terminal Sets and Cost - Linear System, Quadratic Cost

$$
J_0^*(x_0) = \min_{U_0} \qquad x_N' P x_N + \sum_{k=0}^{N-1} x_k' Q x_k + u_k' R u_k \qquad \text{Terminal Cost}
$$

subj. to

$$
x_{k+1} = A x_k + B u_k, \quad k = 0, \dots, N-1
$$

$$
x_k \in \mathcal{X}, \quad u_k \in \mathcal{U}, \quad k = 0, \dots, N-1
$$

$$
x_N \in \mathcal{X}_f \qquad \text{Terminal Constant}
$$

$$
x_0 = x(t)
$$

Choice of Terminal Sets and Cost - Linear System, Quadratic Cost

Design unconstrained LQR control law

$$
F_{\infty} = -(B'P_{\infty}B + R)^{-1}B'P_{\infty}
$$

where P_{∞} is the solution to the discrete-time algebraic Riccati equation:

 $P_{\infty} = A'P_{\infty}A + Q - A'P_{\infty}B(B'P_{\infty}B + R)^{-1}B'P_{\infty}A$

- Choose the terminal weight $P = P_{\infty}$
- Choose the terminal set \mathcal{X}_f to be the maximum invariant set for the closed-loop system $x_{k+1} = (A + BF_{\infty})x_k$:

$$
x_{k+1} = Ax_k + BF \infty(x_k) \in \mathcal{X}_f, \text{ for all } x_k \in \mathcal{X}_f
$$

All state and input constraints are satisfied in \mathcal{X}_f :

$$
\mathcal{X}_f \subseteq \mathcal{X}, \ F_{\infty} x_k \in \mathcal{U}, \quad \text{for all } x_k \in \mathcal{X}_f
$$

Choice of Terminal Sets and Cost - Linear System, Quadratic Cost

- The stage cost is a positive definite function $\mathbf{1}$
- By construction the terminal set is **invariant** under the local control law $\mathbf{2}$ $v = F_{\infty}x$
- **B** Terminal cost is a continuous Lyapunov function in the terminal set \mathcal{X}_f and satisfies:

 $x'_{k+1}Px_{k+1} - x'_{k}Px_k = x'_{k}(-P_{\infty} + A'P_{\infty}A - A'P_{\infty}B(B'P_{\infty}B + R)^{-1}B'P_{\infty}A)x_k$ $=-x_k'Qx_k$

All the Assumptions of the Feasibility and Stability Theorem are verified.

Example: Unstable Linear System

System dynamics:

$$
x_{k+1} = \begin{bmatrix} 1.2 & 1 \\ 0 & 1 \end{bmatrix} x_k + \begin{bmatrix} 1 \\ 0.5 \end{bmatrix} u_k
$$

Constraints:

$$
\mathcal{X} := \{x \mid -50 \le x_1 \le 50, \ -10 \le x_2 \le 10\} = \{x \mid A_x x \le b_x\}
$$

$$
\mathcal{U} := \{u \mid ||u||_{\infty} \le 1\} = \{u \mid A_u u \le b_u\}
$$

Stage cost:

$$
q(x, u) := x' \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} x + u^T u
$$

Horizon: $N = 10$

Example: Closed-loop behavior

Feasibility and Stability Example: Lyapunov Decrease of Optimal Cost 7000_[$J^{*}(x)$ n

Stability of MPC – Remarks

 \Box The terminal set X_f and the terminal cost ensure recursive feasibility and stability of the closed-loop system.

But: the terminal constraint reduces the region of attraction.

(Can extend the horizon to a sufficiently large value to increase the region)

Are terminal sets used in **practice**?

- Generally not...
	- \Box Not well understood by practitioners
	- Requires advanced tools to compute (polyhedral computation or LMI)
- Reduces region of attraction
	- A 'real' controller must provide some input in every circumstance
- Often unnecessary

 $□$ Stable system, long horizon $→$ will be stable and feasible in a (large) neighbourhood of the origin

Choice of Terminal Set and Cost: Summary

- Terminal constraint provides a sufficient condition for stability
- Region of attraction without terminal constraint may be larger than for MPC with terminal constraint but characterization of region of attraction extremely difficult
- $\Box X_f = 0$ simplest choice but small region of attraction for small N
- Solution for linear systems with quadratic cost
- \Box In practice: Enlarge horizon and check stability by sampling
- With larger horizon length N , region of attraction approaches maximum control invariant set

MPC

- Basic Ideas of Predictive Control
- Receding Horizon Control Notation
- **OMPC** Features
- Stability and Invariance of MPC
- **T** Feasibility and Stability
	- **O**Proof for $X_f = 0$
	- **O**General Terminal Sets
	- **O** Example
- Extension to Nonlinear MPC

Extension to Nonlinear MPC

Extension to Nonlinear MPC

Consider the nonlinear system dynamics: $x(t + 1) = g(x(t), u(t))$

$$
J_0^*(x(t)) = \min_{U_0} \qquad p(x_N) + \sum_{k=0}^{N-1} q(x_k, u_k)
$$

subj. to $x_{k+1} = g(x_k, u_k), k = 0, ..., N-1$
 $x_k \in \mathcal{X}, u_k \in \mathcal{U}, k = 0, ..., N-1$
 $x_N \in \mathcal{X}_f$
 $x_0 = x(t)$

- **Q** Presented assumptions on the terminal set and cost did not rely on linearity
- Lyapunov stability is a general framework to analyze stability of nonlinear dynamic systems
- \rightarrow Results can be directly extended to nonlinear systems.

However, computing the sets X_f and function p can be very difficult

Summary:

Finite-horizon MPC may not be stable! Finite-horizon MPC may not satisfy constraints for all time!

An infinite-horizon provides stability and invariance.

- We 'fake' infinite-horizon by forcing the final state to be in an invariant set for which there exists an invariance-inducing controller, whose infinitehorizon cost can be expressed in closed-form.
- These ideas extend to non-linear systems, but the sets are difficult to compute.